READOUT AND CONTROL OF A POWER-RECYCLED INTERFEROMETRIC GRAVITATIONAL WAVE ANTENNA *

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Abstract

Interferometric gravitational wave antennas are based on Michelson interferometers whose sensitivity to small differential length changes has been enhanced by adding multiple coupled optical resonators. The use of optical cavities is essential for reaching the required sensitivity, but sets challenges for the control system which must maintain the cavities near resonance. The goal for the strain sensitivity of the Laser Interferometer Gravitational-wave Observatory (LIGO) is $10^{-21}$ rms, integrated over a 100 Hz bandwidth centered at 150 Hz. We present the major design features of the LIGO length and frequency sensing and control system which will hold the differential length to within $5 \times 10^{-14}$ m of the operating point. We also highlight the restrictions imposed by couplings of noise into the gravitational wave readout signal and the required immunity against them.

1 INTRODUCTION

The interferometric gravitational wave detectors currently under construction by LIGO [1], VIRGO [2], GEO [3] and TAMA [4] are expected to reach strain sensitivity levels of $\sim 10^{-22}/\sqrt{\text{Hz}}$ at 150 Hz over baselines of several hundred meters up to several kilometers [5]. To achieve this sensitivity all of these interferometers implement a Michelson laser interferometer enhanced by multiple coupled optical resonators [6, 7].

LIGO implements a power-recycled Michelson interferometer with Fabry-Perot arm cavities (see Fig. 1). Using optical cavities is essential in reaching the ultimate sensitivity goal but it requires an active electronic feedback system to keep them “on resonance”. The control system must keep the round-trip length of a cavity near an integer multiple of the laser wavelength so that light newly introduced into the cavity interferes constructively with light from previous round-trips. Under these conditions the light inside the cavity builds up and the cavity is said to be on resonance [8]. Attaining high power buildup in the arm cavities also requires that minimal light is allowed to leave the system through the antisymmetric port, so that all the light is sent back in the direction of the laser where it is reflected back into the system by the power recycling mirror. Hence, an additional feedback loop is needed to control the Michelson phase so that the antisymmetric port is set on a dark fringe.

2 ENVIRONMENTAL INFLUENCES

It is important to distinguish low ($< 50$ Hz) and high frequency behaviour of the instrument. The low frequency region is typically dominated by environmental influences many orders of magnitude larger than the designed sensitivity and in many cases also many orders of magnitude larger than what can be tolerated for stable operations. It is the high frequency regime which yields good sensitivity and which is used for detecting gravitational waves. To suppress low frequency disturbances many active feedback control systems are needed to compensate 4 longitudinal [9] and 14 angular [10] degrees-of-freedom in the main interferometer alone. Additional feedback compensation networks are needed to locally damp the suspended mirrors ($13 \times 4$ dofs), to control the mode cleaner (5 dofs) and to control the laser (2 dofs).

For example, seismic motion of the ground [13] is many orders of magnitude larger than the required gravitational wave sensitivity. In LIGO a multi-stage passive seismic isolation stack [11] together with a single-stage pendulum suspension system [12] is used to isolate the optical components from ground vibrations. This system works well for frequencies above $\sim 10$ Hz, but gives no suppression at frequencies of a Hz and below.

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* Work supported by National Science Foundation cooperative agreement PHY-9210038
Figure 1: Schematic view of the optical path in LIGO. The light of a frequency stabilized Nd:YAG laser is passed through a triangular mode cleaner cavity before it is launched into a Michelson interferometer. To stabilize the laser frequency a small fraction of the light is sampled, doubly passed through an acousto-optic modulator (AOM) which serves as a frequency shifter, passed through a Pockels cell and sent to a reference cavity. Using a polarizing beamsplitter (PBS) and quarter-wave plate ($\lambda/4$) the light reflected from the reference cavity is measured by a photodetector to obtain the error signal, $S_{\text{ref}}$, which in turn is used to adjust the laser frequency. The main laser light is passed through a pre-modecleaner (not shown) and two Pockels cells which impose the phase-modulated rf sidebands used to lock the mode cleaner and the Michelson interferometer. The mode cleaner locking signal, $S_{\text{MC}}$, is measured by a photodetector in reflection of the mode cleaner cavity. The light which passes through the mode cleaner is sent through a Faraday isolator (FI) which also serves the purpose—together with a polarizer (P)—to separate out the reflected light signal, $S_{\text{refl}}$. The main interferometer consists of a beamsplitter (BS), two arm cavities each of them formed by an input test mass (ITM) and an end test mass (ETM), and the power recycling mirror (PRM). Additional locking signals are obtained at the antisymmetric port, $S_{\text{anti}}$, and by sampling a small amount of light from inside the power recycling cavity, $S_{\text{prc}}$.

3 FEEDBACK COMPENSATION NETWORK

In order to implement feedback each degree-of-freedom which is under control of the compensation network has to be measurable. LIGO implements the Pound-Drever-Hall reflection locking technique [14] to keep cavities on resonance and a variant of this technique is used to control the angular degrees-of-freedom [15]. These techniques work well near resonance where they behave linearly but have a strong non-linear behaviour far way from resonance giving no or misleading signals. The first step of engaging the feedback compensation network is to catch the system on resonance with a highly sophisticated computer code [16] running on a digital control system.

A schematic view of the length control system for the common mode degrees-of-freedom is shown in Fig. 2. The signal $S_{\text{refl}}$ measuring the common arm length of the interferometer is fed back to a combination of test masses, mode cleaner length and laser frequency to achieve the required laser frequency noise suppression of $< 10^{-6} \text{Hz}/\sqrt{\text{Hz}}$ in the frequency band of interest. To maintain maximum optical power in the system—and thus maximum signal to shot noise ratio—the control system must hold the common cavity length within $< 2 \times 10^{-12} \text{ m rms}$ of its resonance point. A similar but less complicated system is deployed to control the differential degrees-of-freedom. Their the differential arm cavity length has to be held within $< 5 \times 10^{-14} \text{ m rms}$ of its operating point to not pollute the gravitational wave signal with laser frequency noise.

4 CONCLUSIONS

So far LIGO has successfully demonstrated that the interferometer can be locked and kept on resonance for hours. The main goal in the near term is to improve the sensitivity which is still many orders of magnitude away from design, to engage the remaining feedback control paths and to fine-tune servo parameters.
Figure 2: Common mode control system. The mode cleaner error signal, \( S_{MC} \), is split into two paths: the mode cleaner length path (1) feeding back to the position of a mode cleaner mirror, \( L_{MC} \), and the laser path (2) feeding back to the laser frequency, \( f_{laser} \), using the VCO/AOM. The in-phase reflection signal, \( S_{refl} \), of the interferometer (ifo) is split into four paths: the arm cavity path (3) feeding back to the common arm cavity mirror positions, \( L_+ \), the additive offset (ao) path (4) feeding back to the error point of the mode cleaner control system, the mode cleaner length offset path (5) feeding back to the mode cleaner mirror position, \( L_{MC}^{ofs} \), and the tidal path (6) feeding back to the reference cavity length, \( L_{ref} \), using the thermal actuator. The in-phase signal at the power recycling cavity port, \( S_{prec} \), is mostly sensitive to the power recycling cavity length, \( L_+ \), and is feed back to the recycling mirror position (7). The numbers in the feedback paths indicate unity gain frequencies in hertz.

5 REFERENCES


