

**SLAC Summer Institute 2005**

**Gravity Probe B:  
Testing General Relativity  
in Space**

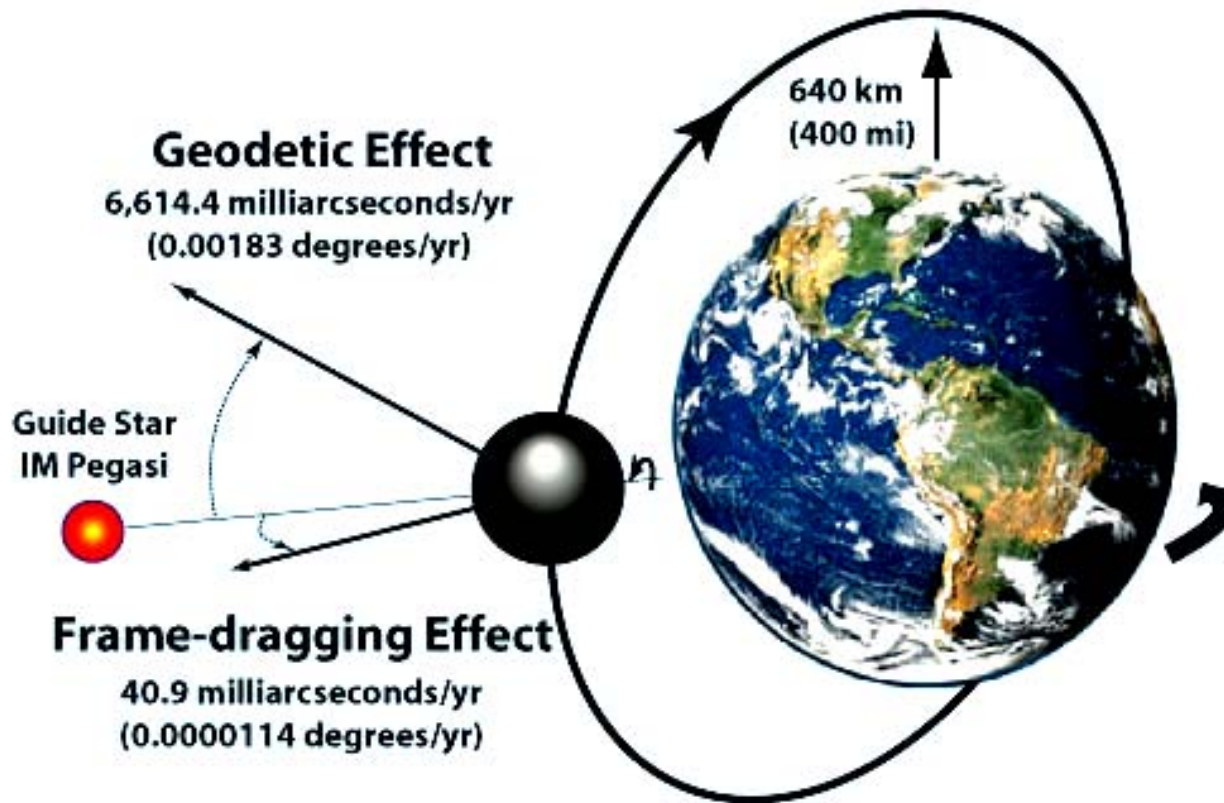
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**July 25, 2005**

**Mission Day 461**



# The Relativity Mission Concept



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- Guide Star and Star Proper Motion Studies**
- Independent Science Analysis**
- Guide Star and Star Proper Motion Studies**
- Helium Ullage Behaviour**
- Gyroscope Read-out Topics**
- Proton Monitor**
- High Precision Homogeneity Measurement of Quartz**

# Fundamental GP-B Requirement

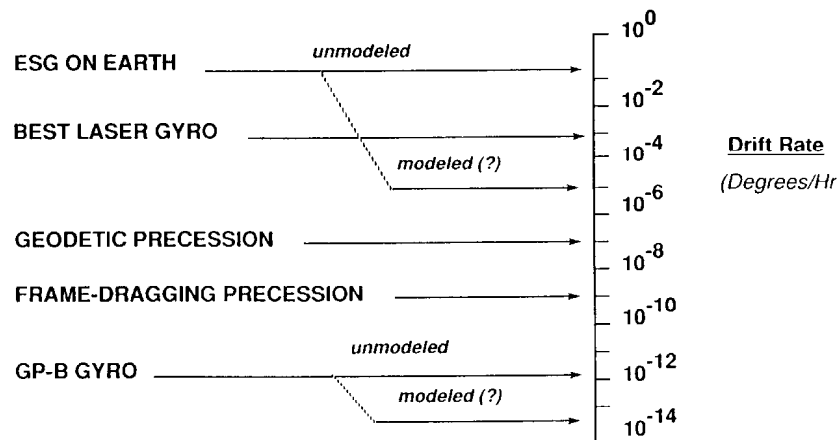
## Gyro Newtonian Drift Rates < GR Effects Under Test

==> no modeling of Gyro spin dynamics or subtraction of Newtonian effects is necessary

Achieved by controlling “near zeros”

- 1) rotor inhomogeneities
- 2) "drag-free"
- 3) rotor asphericity
- 4) magnetic field
- 5) pressure
- 6) electric charge

## Why go to space?



- Gyroscope drift  $\leq 0.05$  marcsec/yr
- Readout error effect  $\leq 0.08$  marcsec/yr
- Guide star proper motion uncertainty  $\leq 0.09$  marcsec/yr

$1 \text{ marcsec/yr} = 3.2 \times 10^{-11} \text{ deg/hr}$

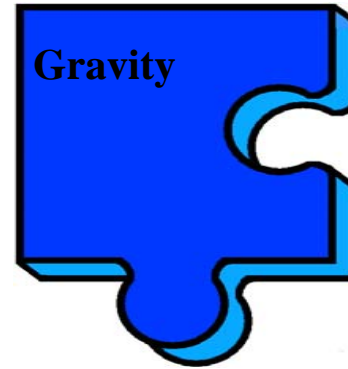
1 marcsec ~ width of a human hair seen from 10 miles

# Why Measure Gravity?

## General Relativity = Present Theory of Gravity

Mathematically Consistent

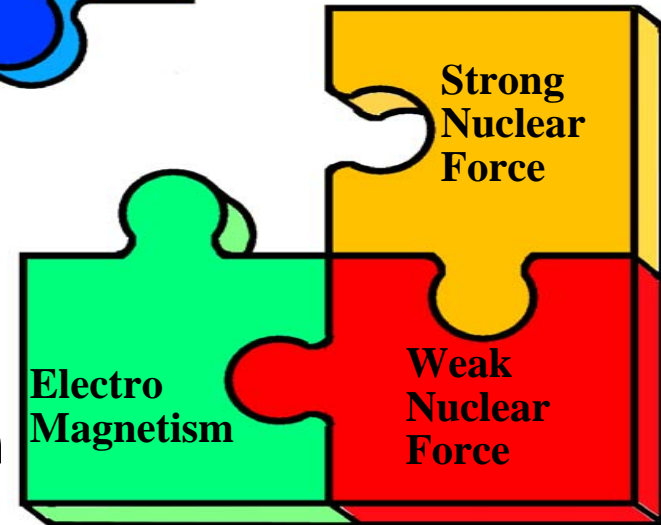
Agrees with Observation (so far)



## Unified Physics ?

Standard Model: Quantum Gauge Theories

GR cannot be quantized



## Partial steps toward Grand Unification

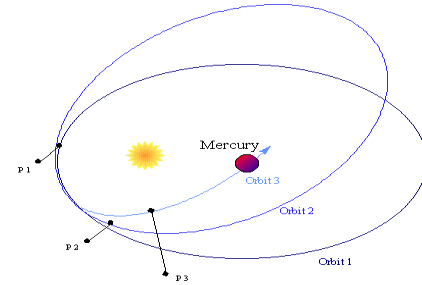
Strings/supersymmetry in early Universe --  
scalar-tensor theory, not Einstein's

Damour - Polyakov -- long range,  
equivalence-violating dilaton

# Einstein's 2 1/2 Tests

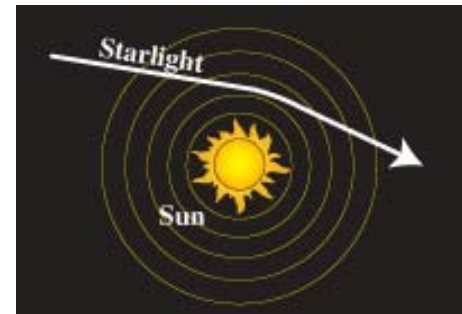
## Perihelion Shift of Mercury

GR resolved 43 arcsec/century discrepancy



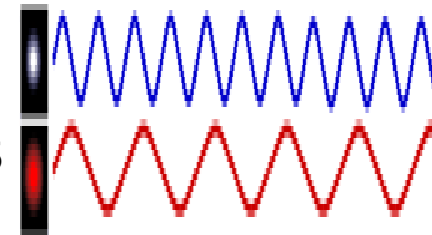
## Deflection of light by the sun

GR correctly predicted 1919 eclipse data  
1.75 arcsec deflection



## Gravitational Redshift -- Test of EP

1960 Pound-Rebka experiment,  $\Delta\nu/\nu=2.5 \times 10^{-15}$   
1976 Vessot-Levine GP-A



Testing GR requires high precision, even the sun is a weak source

# Recent Tests of GR

1968 – Through present

Shapiro Time Delay Viking

Recent Result - Cassini Spacecraft:  $5 \times 10^{-5}$

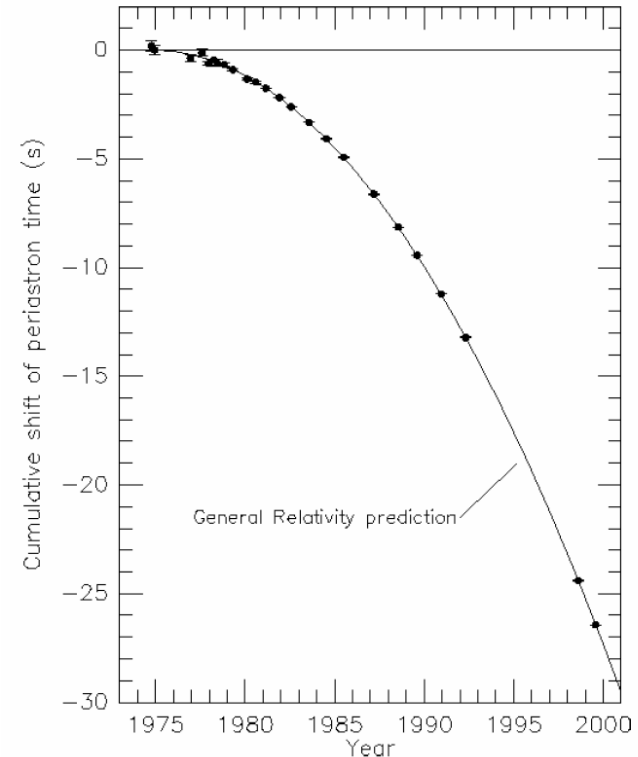
1969 – Through present

Lunar Laser Ranging

EP, Nordtvedt Effect, Geodetic Effect

1974 – Through present

Taylor Hulse Binary Pulsar- Evidence for  
Gravitational radiation



# Tests of General Relativity: Background

Einstein Equivalence Principle (EEP)

Weak EP – Universality of Free Fall

Local Lorentz Invariance

Local Position Invariance

Gravitational energy Gravitates

EEP  $\rightarrow$  metric theory of Gravity

events in spacetime separated by invariant line element

$$ds^2 = g_{\mu\nu} dx^\mu dx^\nu$$

objects in free fall follow geodesics of the metric

Weak Field Limit  $g_{\mu\nu} = \eta_{\mu\nu} + h_{\mu\nu}$        $\eta_{\mu\nu}$  is Minkowski metric

# Tests of General Relativity: Background 2

Einstein Field Equation  $G_{\mu\nu} = R_{\mu\nu} - 1/2 g_{\mu\nu} R = (8\pi G/C^4)T_{\mu\nu}$

“matter tells spacetime how to curve, and curved space tells matter how to move”

No adjustable parameters

– G directly measurable Newtonian gravitational constant

Schwarzschild solution: static, spherically symmetric field of a point mass

– weak field expansion to first order

$$ds^2 = (1-2GM/C^2R)C^2dt^2 - (1+2GM/C^2R)dr^2$$

$$g_{00} = -(1-2\Phi/C^2) \quad \Phi = GM/RC^2$$

At surface of a proton  $\Phi = 10^{-39}$   
At surface of a proton  $\Phi = 10^{-9}$   
At surface of a proton  $\Phi = 10^{-6}$

For solar system tests spacetime distortions due to gravity are small

# Metric Outside Rotating Earth

The metric outside the rotating earth to lowest non trivial order in  $\Phi$  and in  $v$  - the relevant motions involved is given by:

$$g_{00} = (1 - 2\Phi/C^2)$$

$$g_{ij} = -\delta_{ij} (1 + 2\Phi/C^2)$$

$$g_{i0} = \zeta_i \quad \text{where} \quad \zeta = 2GI_e/C^2R^3 (\mathbf{R} \times \boldsymbol{\omega}_e)$$

$I_e$  and  $\boldsymbol{\omega}_e$  are the moment of inertia and rotation velocity of the earth

# Metric Outside Rotating Earth 2

Plausibility argument and interpretation

Working to first order (and setting  $C=1$ ) we start with the line element for a stationary point mass  $ds^2 = (1-2GM/R)dt^2 - (1+2GM/R)dr^2$

Generalize to moving point mass by transforming to moving system w/ 1st order Lorentz xform  $t = t-vx$   $x_r = x-vt$ . (for motion in x direction)  $\longrightarrow ds^2 = (1-2GM/R)dt^2 - (1+2GM/R)dr^2 + 4m/r v dx dt$

So for motion in general direction:

$$ds^2 = (1-2GM/R)dt^2 - (1+2GM/R)dr^2 + 4m/r \mathbf{v} \cdot \mathbf{dr} dt$$

Since the theory is linear to this order, can superpose distribution of point masses to get  $ds^2 = (1-2\Phi)dt^2 - (1+2\Phi)dr^2 + \mathbf{h} \cdot \mathbf{dr} dt$  where,

$$\Phi(r) = G \int \frac{\rho dr'}{r'} \quad h(r) = 4G \int \frac{\rho v dr'}{r'}$$

Thus  $\Phi(r)$  and  $\mathbf{h}(r)$  are scalar and vector potentials

in analogy with EM, coulombs law + Special Relativity yields magnetism

# Effects of Gravitation on Spin

Particle in Free Fall       $U^\alpha = dx^\alpha/d\tau =$  four velocity       $dU^\alpha/d\tau = 0$

$S_\alpha =$  four spin       $dS_\alpha/d\tau = 0$  in Special Rel

$S_\alpha = \{\mathbf{S}, 0\}$  in rest frame

$S_\alpha U^\alpha = 0$  in all frames

To go from Special Relativity to General Relativity make equations covariant  
—> derivatives go over to:

$$dU^\mu/d\tau + \Gamma^\mu_{\nu\lambda} U^\nu U^\lambda = 0$$

$$dS_\alpha/d\tau - \Gamma^\mu_{\alpha\nu} U^\nu S_\mu = 0 \quad \text{Parallel transport}$$

$$\Gamma^\sigma_{\lambda\mu} = 1/2 g^{\nu\sigma} [ dg_{\mu\nu}/dx^\lambda + dg_{\lambda\nu}/dx^\mu - dg_{\mu\lambda}/dx^\nu ] \quad \text{Affine Connection}$$

# Precession of a Gyro in Earth Orbit

In orbit  $\rightarrow$  no forces on gyro

Therefore Four spin will undergo Parallel transport

$$dS_\alpha/d\tau - \Gamma^\mu_{\alpha\nu} S_\mu dx^\nu/d\tau = 0$$

$$S_\mu dx^\mu/d\tau = 0 \rightarrow S_0 = -V^i S_i$$

So can eliminate  $S_0$  to get:

$$dS_i/dt = \Gamma^j_{i0} S_j - \Gamma^0_{k0} V^k S_i + \Gamma^j_{ik} V^k S_j - \Gamma^0_{ik} V^k V^j S_j$$

Define precession vector  $\Omega$  by  $d\mathbf{S}/dt = \Omega \times \mathbf{S}$

Then using  $g_{\mu\nu}$  of earth to calculate  $\Gamma$ 's and eliminating terms which vanish for periodic orbits one finds:

$$\bar{\Omega} = \bar{\Omega}_G + \bar{\Omega}_{FD} = \frac{3GM}{2c^2 R^3} (\bar{R} \times \bar{v}) + \frac{GI}{c^2 R^3} \left[ \frac{3\bar{R}}{R^2} (\bar{\omega}_e \cdot \bar{R}) - \bar{\omega}_e \right]$$

Where  $R$  is the vector from the earth center to the gyro,  $v$  is the gyro velocity

# Precession of a Gyro in Earth Orbit 2

The first term, the geodetic precession, also called the deSitter effect, is caused by the interaction of the mass of the earth with the gyro frame and the distortion of spacetime by the mass of the earth and depends on the orbital velocity of the gyro

$$\bar{\Omega}_G = \frac{3GM}{2c^2 R^3} (\bar{R} \times \bar{v})$$

The latter term, the Frame Dragging or Lense-Thirring term depends on the angular momentum of the earth. The earth is the prime solar system body for the source since it's angular momentum is well known.

$$\bar{\Omega}_{FD} = \frac{GI}{c^2 R^3} \left[ \frac{3\bar{R}}{R^2} (\bar{\omega}_e \cdot \bar{R}) - \bar{\omega}_e \right]$$

Since its an interaction between spin angular momenta of the earth and gyro this term is sometimes referred to as 'hyperfine interaction', in analogy with atomic physics. But does not depend on magnitude of gyro spin.

Any vector that points a fixed direction in the local inertial frame will precess in frame of distant stars.

# Precession of a Gyro in Earth Orbit 3

For a polar orbit the Frame dragging and geodetic precession are perpendicular and can therefore be resolved independently

Both terms are maximized by taking  $R_{\text{orbit}}$  close to  $R_{\text{earth}}$

For 642km altitude and gyro spin near the equatorial plane

The magnitude of the effects averaged over an orbit are

Geodetic  $\Omega_G = 6.6$  arcsec/year    Frame Dragging  $\Omega_{\text{FD}} = 0.041$  arcsec/year

GP-B goal is to test these to parts in  $10^5$  and to better than 1% resp.

Leading Corrections are due to:

Geodetic effect due to the sun,

$$\Omega_G^s = 19 \text{ marcsec/year}$$

(in plane of the ecliptic)

Oblateness correction of Geodetic Effect,

$$\Omega_G^{J^2} = -7 \text{ marcsec/year}$$

(parallel to  $\Omega_G$ )

Frame dragging due to the sun and effects from the moon are negligible

# Parameterized Post-Newtonian Formalism

Eddington, Nordtvedt, Will

Useful parameterization for a range of metric theories of gravity

Low velocity, weak field limit  
Solar system

Metrics of various theories similar form

Expansion about flat (Minkowski) space

Gravitational potentials provide deviation from flat space

Various theories provide different coefficients

Parameter	Meaning	GR value
$\gamma$	Space curvature	1
$\beta$	Non-linearity in gravity superposition	1
$\xi$	Preferred location	0
$\alpha_{1-3}$	Preferred frame	0
$\Gamma_{1-4}$	Conservation of momentum	0

$$\bar{\Omega}_G = \left( \gamma + \frac{1}{2} \right) \frac{GM}{c^2 R^3} (\bar{R} \times \bar{v})$$

$$\bar{\Omega}_{FD} = \left( \gamma + 1 + \frac{\alpha_1}{4} \right) \frac{GI}{2c^2 R^3} \left[ \frac{3\bar{R}}{R^2} (\bar{\omega}_e \cdot \bar{R}) - \bar{\omega}_e \right]$$

Caveat: Danger in viewing all GR tests in terms of PPN parameter limits. Metric theories that fit within PPN framework span only a part of the space of possible alternative theories. Most recent attempts at unification of the forces of nature require non-metric revisions of GR.

# Significance of Frame Dragging

In GR gravitational effects are due to the geometry of spacetime.

A non-rotating body produces a static spacetime curvature due to Mass. A rotating body produces an additional dragging of external inertial frames via the body's angular momentum, analogous to the body being surrounded by a viscous fluid. The instantaneous direction change of the gyro over the poles is in opposite direction to that when the gyro is above the equator. This is just what a test "straw" oriented perpendicular to the immersed in a fluid would do.

Frame Dragging is conceptually related to Mach's principle: Inertial forces arise from accelerations and rotations with resp. to total mass of the universe. Under this interpretation the gyro spin direction reaches a compromise between following the distant stars and the rotating earth.

Lense and Thirring in 1918 considered the metric inside a rigidly rotating hollow sphere. In 1966 it was shown that as the shell's gravitational radius,  $2GM/c^2$ , approaches its physical radius, the precession of a gyro at the center approaches the angular velocity of the shell. That is, the gyro spin becomes locked to the rotating distant matter. Water would climb up the sides of a non-rotating bucket at the center of the shell.

# Gravitoelectric / Gravitomagnetic Viewpoint

Space-time metric	Newtonian analog	EM analog	Gravito EM analog	Rotational effect
$g_{00}$	$\phi$	$V$	$E_g$	$1/3\Omega_G$
$g_{0i}$	No analog	$A_i$	$B_g$	$\Omega_{FD}$
$g_{ij}$	No analog	No analog	No analog (space curvature)	$2/3\Omega_G$

This view point arises in analogy with electromagnetism or from treating GR as a spin 2 field theory to lowest order. Static matter generates a gravito-electric potential and space curvature. Rotation produces a gravito-magnetic dipole field. Frame dragging arises from the interaction of the gyro spin with a vector potential  $g_{0i}$ .

A moving mass (mass current) generates a gravito-magnetic field; it is related to gravito-electrostatics through Lorentz xform (see slide 11). However, a gravito-magnetic field of a rotating mass cannot be derived by rotation of coordinates from the static field of a non-rotating mass. Frame Dragging as tested by GP-B does not have a trivial relationship to static gravitation.

# Evidence of Frame Dragging Effects

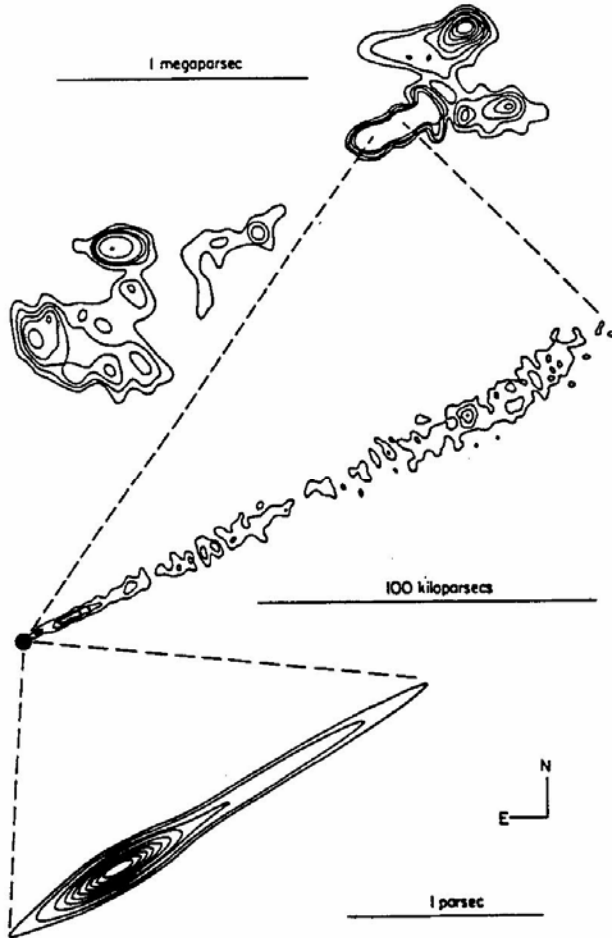
Within GR, frame dragging should produce a precession of the orbital plane of a body in earth orbit (the original prediction of Lense and Thirring) and a precession of the pericenter of the orbit. Ciufolini et al. have analyzed laser ranging data of the LAGEOS I and LAGEOS II satellites and claim a confirmation of FD effects to a 25% level. Results are controversial due to the need to subtract out Newtonian perturbations that are  $> 10^7$  times larger than the GR predictions, as well as understanding non-gravitational disturbances due to radiation pressure and thermal effects.

In the Taylor-Hulse binary pulsar B1913+16, uncertainties in the constituent properties are too large to be used for precision tests of gravitomagnetic effects. A pulsar orbiting a fast rotating black hole could provide an appropriate test system; no definitive candidates are yet known.

Some effects of frame dragging by translational motion are present (in combination with other effects) in Lunar Laser ranging tests of the Nordtvedt effect.

Within the PPN framework, Shapiro time delay and LLR measurements place tight limits on the  $\gamma$  and  $\alpha_1$  parameters.

# Frame-Dragging Evidence in Astrophysics



Observed jets from Galaxy NGC 6251  
Power output  $10^{38}$  watts!  
(A trillion suns)

Compact object holds jet direction aligned millions of years. Plausible cause of the alignment is the effect of frame dragging from a Supermassive Black Hole. “Frame dragging imprints the angular momentum of the source on the distant spacetime.” Evidence for FD but does not allow for quantitative measurement.

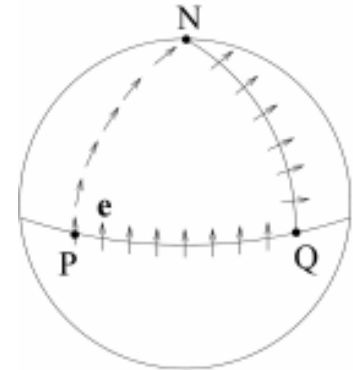
**Radio Map of  
Galaxy NGC 6251**

# Significance of Geodetic Precession

The Geodetic effect is a combination of precession due the gravito- electric field and precession due curved spacetime.

The gravitoelectric part, 1/3 of the total effect, is due to special relativistic corrections to the motion of the gyro in the  $g_{00}$  potential.

The other 2/3 of the precession is due the curvature of space around the earth. Change in vector orientation under parallel transport is a hallmark of curvature as seen in 2 dimensional example. Transporting from points Q-P-N-Q the vector is rotated.



For the GP-B case the curvature is due to the mass of the earth distorting the space around it. This produces an orbit circumference smaller than what one would calculate using  $2\pi r$  in flat space, smaller by about 1 inch over 27000 miles.

Several alternative theories, such as Damour-Nordtvedt "attractor mechanism" tensor-scalar theories of gravity, predict deviations from zero of the PPN parameter  $\gamma$  of up to  $\gamma-1 \leq 4 \times 10^{-5}$  )

# Evidence of Geodetic Effects

For The Earth-Moon system can be considered as a "gyroscope", with its axis perpendicular to the orbital plane. The predicted precession, first calculated by de Sitter, is about 19 marcseconds per year. This effect has been measured to about 0.7 percent using Lunar laser ranging data.

Within the PPN framework, light deflection and Shapiro time delay measurements around the sun set limits on the curvature parameter  $\gamma$ .

The current limits established by telemetry data from the Cassini spacecraft are a few parts in  $10^5$ .

# Conclusion

**GP-B will provide Two Direct Measurements of GR**

## **Frame Dragging**

**Goal: measure to better than 1%**

*First direct measurement of gravitomagnetism associated with rotation*

## **Geodetic Effect**

**Goal: measure to  $\leq 2 \times 10^{-5}$**

*Measurement in region  $10^{-4} > \gamma > 10^{-7}$  where scalar tensor theories predict deviations from GR*

# References

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